

ESA M7 missions candidates

To be launch in ~2037 - 2039

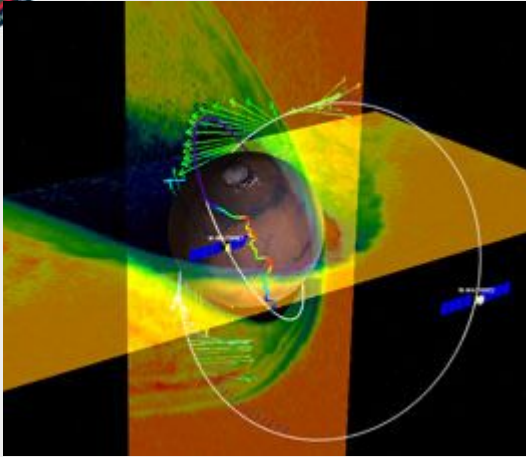
David Píša (Co-PI of M-MATISSE COMPASS)
Institute of Atmospheric Physics CAS



**DEPARTMENT
OF SPACE PHYSICS**
INSTITUTE OF ATMOSPHERIC PHYSICS
CZECH ACADEMY OF SCIENCES



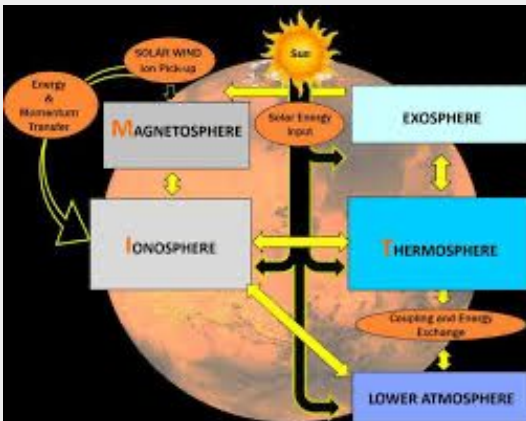
M-MATISSE - Mars Magnetosphere ATmosphere Ionosphere and Space-weather Science



A concept of **two identical spacecraft** (Henri & Marguerite) orbiting around Mars

An unique constellation for studying **Solar wind-Magnetosphere-Ionosphere** coupling at Mars

Both M-MATISSE spacecraft have 6 scientific instruments.



Henri has periapsis below 270 km and an apoapsis of 3000 km with an inclination of 60°

Marguerite has an inclination of 60° , a periapsis below 270 km and an apoapsis of 10,000 km



COMPASS - Wave Analyzer and Processing Unit

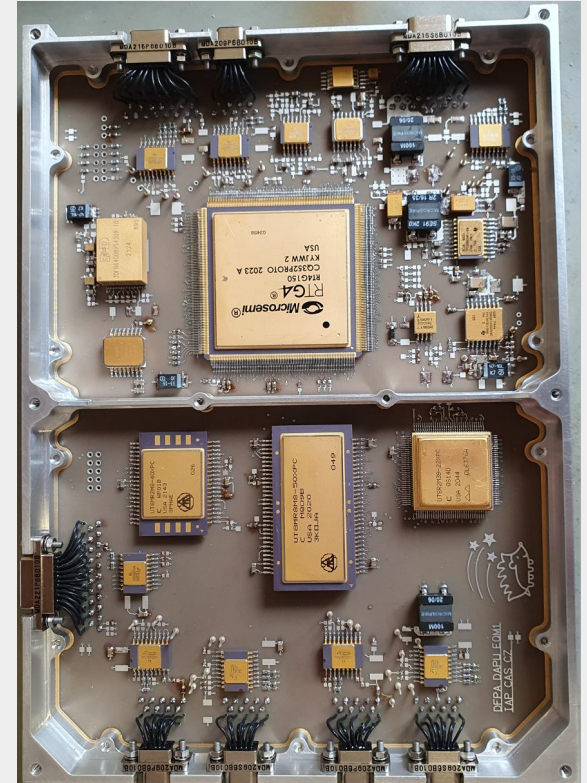


COMPASS: Combined Magnetic and Plasma Sensor Suite

WAPU serves as a common Data Processing Unit

- implements the single common digital interface of COMPASS to the spacecraft
- performs data processing, compression and buffering
- cold redundant unit with more instrument interfaces, based on fast RTG4 FPGA
- Provides on-board classification of wave & dust impacts
- Design heritage – DFP/DAPU (Comet Interceptor)
- Phase B term (BBM/EM AIT) covers 06/2026-12/2028
- **Possible CZ industrial involvement**
 - ECSS-qualified PCB assembly (Phase C/D)

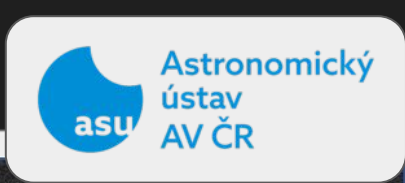
Designed and developed by **Institute of Atmospheric Physics, CAS**
(Co-PI – David Píša, Technical Lead – Jan Souček)



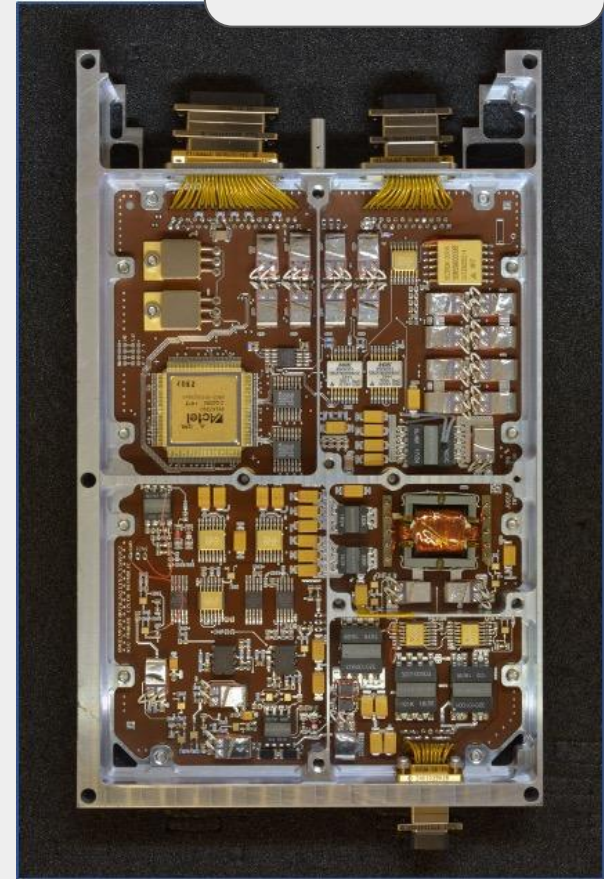
C-I DAPU EQM1



COMPASS - Low Voltage Power Supply (LVPS)



- Designed and developed by **Astronomical Institute, CAS**
(Co-PI – Štěpán Štverák, Technical Lead – Jaroslav Laifr)
- Provides prime power interface of all COMPASS sub-systems to main S/C bus and galvanic isolation of the instrument secondary ground
- Realized as a cold redundant system consisting of two identical and swappable units
- Based on discrete approach of voltage-fed Push Pull DC/DC converters to supply secondary output voltages for the COMPASS subsystems (+3.7V, $\pm 12V$) with total output power of 27 W
- Implements telemetry and telecommand engine in FPGA with LVDS serial link
- Design heritage – LVPS/RPW (Solar Orbiter), LVPS/RPWI (JUICE), GIM/WFI (NewAthena)
- Phase B term (BBM/EM AIT) covers 06/2026-12/2028
- **Anticipated CZ industrial support**
 - Product & Quality Assurance
 - VHDL/FPGA
 - ECSS-qualified PCB assembly (Phase C/D)



SP@M (Solar Particles @ Mars) instrument



Faculty of Mathematics
and Physics
Charles University

Measure energy spectrum & angular distribution of solar energetic electrons (30 keV – 1 MeV)

Time resolution 1 s in burst mode (32 s in nominal mode)

Two identical sensors (SP@M1, SP@M2) on both orbiters, two detection units per sensor (covering opposite directions)

Single-layer solid-state detectors

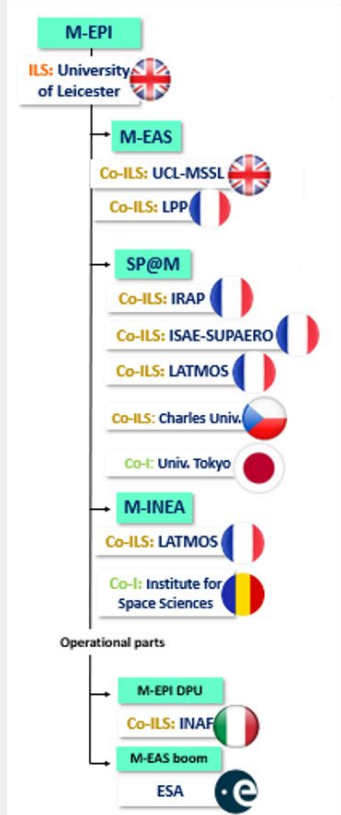
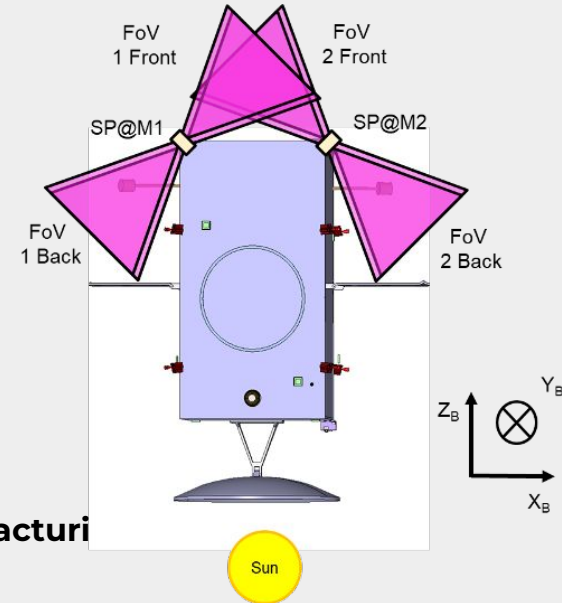
Data acquisition and processing:

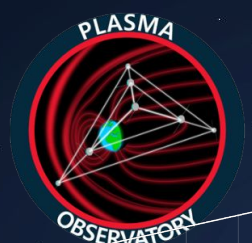
- 2 fast dual-channel ADCs (125 MSps)
- reprogrammable FPGA, digital pulse-processing electronics
- another FPGA (or fast MCU) for instrument command & control
- dual Spacewire interface to the M-EPI DPU (nominal & redundant)

Seeking assistance in:

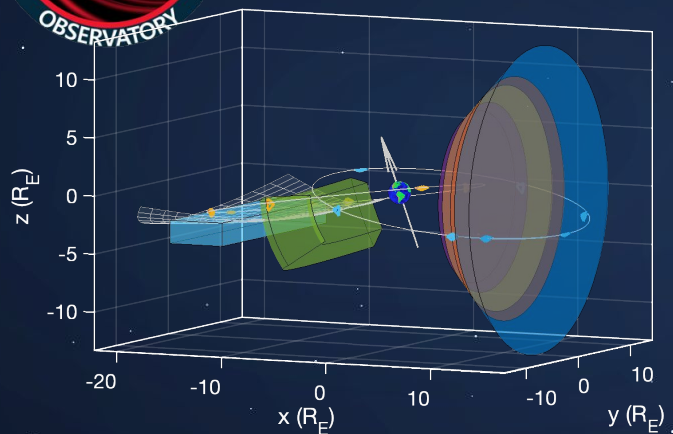
- **FPGA (I/F) board design, construction & manufacturing flight units)**
- **Board programming & data processing**

Czech Lead: Doc. Lubomír Přeč (lubomir.prech@mff.cuni.cz)





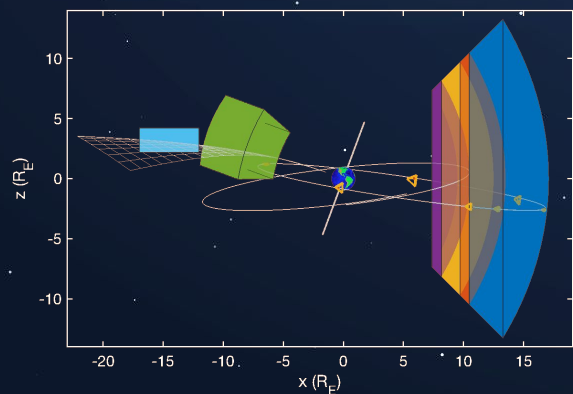
Plasma Observatory

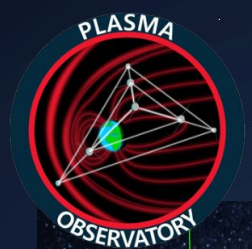


A unique concept for studying wide range of spatio-temporal scales of plasma processes in near-Earth space

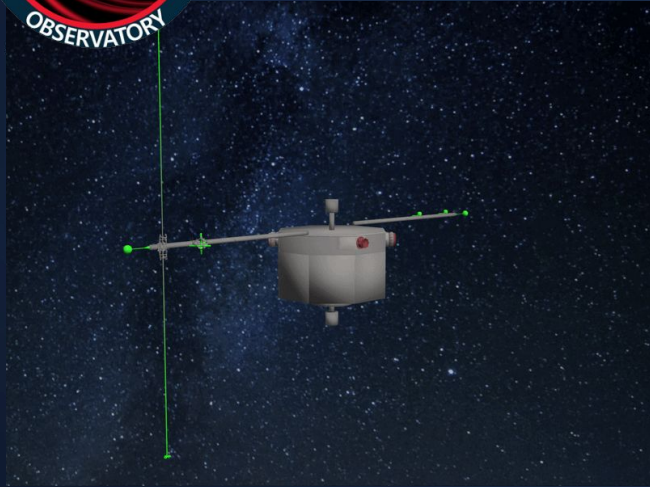
Constellation of 7 small spacecraft

- 200 kg dry mass
- apogee 15 Earth radii





Plasma Observatory



Institute of Atmospheric Physics will develop a wave analyzer - instrument to study plasma waves

Wave analyzer developed in collaboration with a French partner organization (LPC2E, Orleans).

- IAP will develop flight software and firmware.
- Hardware contribution is TBC, depending on CNES funding.
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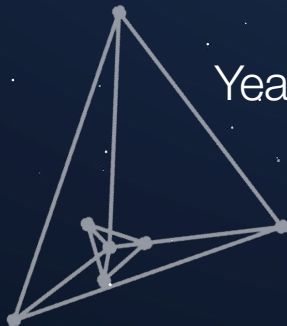
Possible Czech industry involvement:

- **software / firmware independent validation**
- **Flight HW manufacturing and testing (TBC)**

Year 1



Year 2



Year 3

